

1 Optimization of functionally graded materials to make stress 2 concentration vanish in a plate with circular hole

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7 **Abstract:** This paper is devoted to the minimization of the stress concentration factor in
8 infinite plates with circular hole made of functionally graded materials and subjected to a
9 far-field uniform uniaxial tension. Despite the vast literature on the versatility of these
10 materials, the novelty of the results is that the optimal material distribution is not limited
11 to prefixed laws, as in many works available in the literature. It is assumed to be an
12 unknown piecewise constant function, thus aiming to derive the material distribution by
13 exploiting, at best, the inhomogeneity concept associated with functionally graded
14 materials. After a brief review of the governing equations, the motivation, the statement
15 and the mathematical formulation of the optimization problem are given under the
16 hypothesis of axisymmetric material distribution. Still, the problem could not be solved
17 analytically, therefore a direct transcription approach by the aid of finite difference method
18 has been followed to convert it into a nonlinear programming problem, whose solution has
19 been obtained numerically by dedicated gradient-based solvers. Numerical solutions are
20 reported in graphical forms, thoroughly discussed and validated by means of the finite
21 element method. The developed numerical approach yields a material inhomogeneity
22 obeying a sigmoid-like function and a uniform hoop stress along the radial direction, thus
23 making the stress concentration factor at the rim of the circular hole vanish.

24 **Keywords:** Functionally graded materials; stress concentration factor; direct transcription;
25 optimization; nonlinear programming; plates.

27 1. Introduction

28 The study of the stress concentration in panels due to the presence of circular holes
29 constitutes one of the classic problems in mechanics. It is known that if the panel is
30 infinitely large and made of a homogenous, linearly elastic and isotropic material and
31 subjected to a uniform uniaxial tension, then the stress concentration factor (hereinafter

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32 abbreviated by SCF) is identically 3. In literature, this result is commonly referred to as the
33 Kirsch solution, named after the German engineer who first described the elastic stresses
34 around the hole [1]. Since then, engineers and researchers have been interested in reducing
35 such a factor by abandoning the aforementioned isotropic and homogeneity assumptions
36 and the shape of the geometrical discontinuity (see, e.g., [2,3] for an exhaustive literature
37 review on various analytical methods).

38 The adoption of functionally graded materials has propounded its application to
39 numerous mechanical and geotechnical models [4-6], where the microstructure was
40 allowed to vary along one or several directions by employing isotropic, orthotropic or even
41 anisotropic constituent materials (see, e.g., [7-9]). Among all, the stress analysis of
42 functionally graded panels with holes has been investigated. Several analytical and
43 numerical efforts have been carried out aiming at reducing the stress concentration by
44 taking advantage of different inhomogeneity models. For instance, the effect of the material
45 inhomogeneity on the SCF due to circular and elliptic holes are predicted in [10] and [11],
46 respectively, both by means of the finite element method. In particular, Young's modulus
47 has been allowed to vary spatially. Authors have shown that a reduction in the SCF can be
48 obtained by properly choosing the tuning parameters of the heterogeneity factors
49 associated with the property variations (e.g., the exponents in the power- and exponential
50 laws). In [12], the SCF around a circular hole in an infinite plate subjected to uniform
51 biaxial tension and pure shear is analytically solved by exploiting Frobenius series. Closed-
52 form solutions are derived for an exponential variation of Young's modulus along the
53 radius. By dividing the functionally graded plate into a series of piecewise homogeneous
54 radial layers, Ref. [13,14] report the SCFs due to circular holes and under constant loads
55 by means of Muskhelishvili method of the complex variable functions. In [15], closed-
56 form solutions for the SCF at a circular hole in functionally graded panels subject to a
57 uniform far-field tensile traction are derived by using hypergeometric functions and
58 Frobenius series. Authors show that the SCF at the circular hole can be considerably
59 reduced by appropriately grading the mechanical properties along the radial direction. The
60 elastic response of a functionally graded annular ring inserted in a hole of a homogeneous
61 plate is derived analytically in [16,17] under different far-field loading conditions. All the
62 aforementioned works report a considerable stress concentration reduction only when the
63 Young's modulus progressively increases away from the hole. Moreover, it is observed

64 that the variation of the Poisson's ratio on the stress distribution in the plate is negligible
65 [15-17].

66 The aforementioned considerations bring into mind the possibility of exploiting
67 optimization theory to enhance the elastic performance of such structures. Many solutions
68 have been proposed to different problems [18,19], some of which are capable of handling
69 only one-dimensional material distribution with one-dimensional geometry and simple
70 loads, while others can tackle more sophisticated problems. Interesting results in terms of
71 stress reduction have been achieved when considering models such as beams, cylindrical
72 shells, rotating disks, pressure vessels and plates (see, e.g., [20-34]), however by imposing
73 prefixed laws for the variation of mechanical properties. In this way, the optimization
74 problems reduce to the search for the heterogeneity factors associated with functional
75 models describing these property variations. On the other hand, other works dealt with the
76 search for the best material distribution to enhance the elastic stress performance *without*
77 prefixing the functional model. Some of these are developed within an analytical tailoring
78 framework [35,36], whereas others rely on phase-field and topology optimization [37,38]
79 or exploit principles from the optimal control theory [39-41]. As far as infinite plates with
80 a circular hole are concerned, the overwhelming research works impose the Young's
81 modulus a priori to forecast the stress concentration near the hole. Only in Ref. [35], an
82 analytical solution is proposed for the cylinder under pressure, whose validity can
83 equivalently hold for the case of biaxially loaded plates. In the uniaxial load case, to the
84 extent of the authors' knowledge, Ref. [42] is the only work where the unknown Young's
85 modulus distribution is sought in plates with different holes and cutouts, in which enhanced
86 stress results have been obtained by developing an evolutionary algorithm combined with
87 the finite element method. It is worth noting that the iteration process for updating the
88 Young's modulus in each element was governed by a power-law function of local and
89 global stress measures. The stiffness was thus reduced only in the elements whose stresses
90 were higher than an imposed threshold. Although this rule-of-thumb stiffness modification
91 led to enhanced SCFs, we strongly believe that optimal solutions can be achieved if the
92 stiffness optimization is carried out in a more global sense. Accordingly, the objective of
93 the present article is to seek the Young's modulus distribution around the circular hole such
94 that the hoop stress reaches its minimum value along prescribed directions.

95 The article is organized as follows. Section 2 recalls the governing equations for the
 96 plane stresses in linearly elastic, isotropic and inhomogeneous plates. Section 3 aims at
 97 presenting the motivation of the work as well as the formulation of the optimization
 98 problem. Section 4 illustrates the direct transcription approach as a numerical procedure to
 99 convert the optimization problem into a nonlinear programming problem, whose solution
 100 has been computed by resorting to a solver available in the literature. The optimal solution
 101 of a study case, its validation by a finite element model and its discussion are shown in
 102 Section 5 and conclusions are drawn in Section 6.

103 **2. Governing equations**

104 Consider a linearly elastic, isotropic and functionally graded infinite plate with a
 105 circular hole of radius a . Let the thickness of the plate be sufficiently small to the point
 106 that the stress state is two-dimensional (plane-stress condition). Let the plate be subject to
 107 a far-field uniaxial traction σ_0 , as shown in Figure 1a, where the generic point P is described
 108 by the polar coordinate system (r, θ) , whose origin is at the center of the circular hole, and
 109 MN denotes the vertical line associated with the polar angle $\theta = \pi/2$. Moreover, let the
 110 inhomogeneity be described by the radial variation of the volume fraction $V(r)$ of one of
 111 the two constituents of the functionally graded material (e.g. material #2), which in turn
 112 are linked to the effective Young's modulus $E(r)$ by the well-known rule of mixture

$$E(r) = \tilde{E}_1(1 - V(r)) + \tilde{E}_2V(r), \quad (1)$$

113 where \tilde{E}_1 and \tilde{E}_2 denote the Young's moduli of the constituents (e.g., metallic and ceramic
 114 materials), while the Poisson's ratio ν is assumed to be constant and not affected by the
 115 volume fraction. It is worthwhile to note Eq. (1) is adopted in this study since it can be
 116 considered as the simplest homogenization technique among the several approaches in
 117 micromechanics [43].

118 **2.1. Equilibrium, constitutive and compatibility equations**

119 Next, equations describing the mechanical behavior of the plate are listed. In the
 120 absence of body forces, the equilibrium equations read [44]

$$\frac{\partial \sigma_r(r, \theta)}{\partial r} + \frac{1}{r} \frac{\partial \sigma_{r\theta}(r, \theta)}{\partial \theta} + \frac{\sigma_r(r, \theta) - \sigma_\theta(r, \theta)}{r} = 0, \quad (2a,b)$$

$$\frac{\partial \sigma_{r\theta}(r, \theta)}{\partial r} + \frac{1}{r} \frac{\partial \sigma_{\theta}(r, \theta)}{\partial \theta} + \frac{2}{r} \sigma_{r\theta}(r, \theta) = 0 ,$$

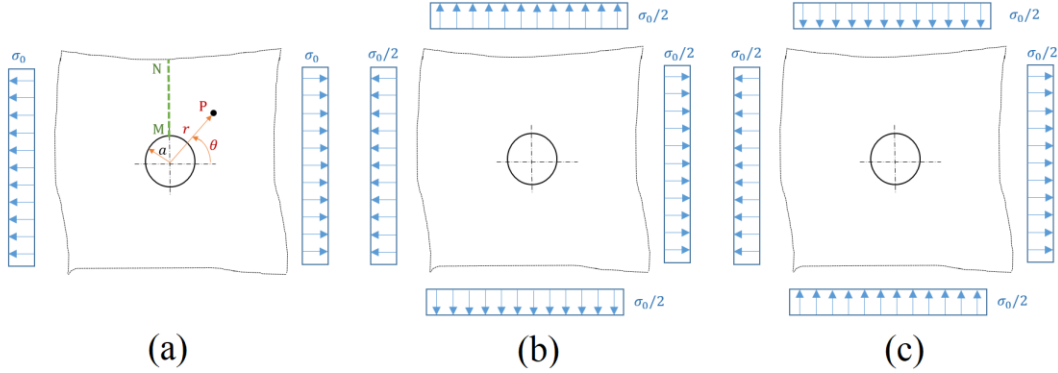


Fig. 1: A schematic representation of (a) an infinite plate with a circular hole subject to a far-field uniaxial traction and its split into (b) uniform biaxial and (c) pure shear sub-problems.

121 where σ_r , σ_{θ} and $\sigma_{r\theta}$ are the radial, hoop and shear stresses, respectively, all functions of
 122 the radial r and circumferential θ coordinates. The elastic stresses are related to the
 123 corresponding strains by the plane-stress constitutive equations, namely [44]

$$\begin{aligned} E(r) \varepsilon_r(r, \theta) &= \sigma_r(r, \theta) - \nu \sigma_{\theta}(r, \theta) , \\ E(r) \varepsilon_{\theta}(r, \theta) &= \sigma_{\theta}(r, \theta) - \nu \sigma_r(r, \theta) , \\ E(r) \varepsilon_{r\theta}(r, \theta) &= 2(1 + \nu) \sigma_{r\theta}(r, \theta) , \end{aligned} \tag{3a-c}$$

124 where ε_r , ε_{θ} and $\varepsilon_{r\theta}$ are the radial, hoop and shear strains, respectively, which obey the
 125 following compatibility equation [44]

$$\frac{\partial^2 \varepsilon_{\theta}}{\partial r^2} + \frac{1}{r^2} \frac{\partial^2 \varepsilon_r}{\partial \theta^2} + \frac{2}{r} \frac{\partial \varepsilon_{\theta}}{\partial r} - \frac{1}{r} \frac{\partial \varepsilon_r}{\partial \theta} = \frac{1}{r} \frac{\partial^2 \varepsilon_{r\theta}}{\partial r \partial \theta} + \frac{1}{r^2} \frac{\partial \varepsilon_{r\theta}}{\partial \theta} . \tag{4}$$

126 2.2. Superposition of stresses

127 Due to the linearity hypothesis, if the elastic problem is split into two sub-problems,
 128 namely the biaxial problem (Figure 1b) and the pure shear problem (Figure 1c), the
 129 superposition of their solutions leads to the solution of the original one. In other words,
 130 letting superscripts “*bx*” and “*ps*” denote respectively the uniform biaxial and pure shear
 131 terms, stresses can be written as

$$\begin{aligned} \sigma_r(r, \theta) &= \sigma_r^{bx}(r) + \sigma_r^{ps}(r, \theta) , \\ \sigma_{\theta}(r, \theta) &= \sigma_{\theta}^{bx}(r) + \sigma_{\theta}^{ps}(r, \theta) , \\ \sigma_{r\theta}(r, \theta) &= \sigma_{r\theta}^{ps}(r, \theta) , \end{aligned} \tag{5a-c}$$

132 where it is emphasized that stresses for the biaxial problem depend on the radial coordinate
 133 only, since the geometry of the problem, the assumed nature of the inhomogeneity and the
 134 far-field loading are axisymmetric (and therefore $\sigma_{r\theta}^{bx}$ is identically zero). Substitution of
 135 the constitutive relations (3a-c) into the compatibility equation (4) yields the following
 136 boundary-value problem for the radial stress

$$\begin{aligned} \mathcal{BX}(\sigma_r^{bx}(r)) &= 0, & a \leq r < \infty \\ \sigma_r^{bx}(a) &= 0, \\ \lim_{r \rightarrow \infty} \sigma_r^{bx}(r) &= \frac{\sigma_0}{2}, \end{aligned} \tag{6a-c}$$

137 where the differential operator $\mathcal{BX}(\cdot)$ is given by $\frac{d^2(\cdot)}{dr^2} + \alpha^{bx}(r) \frac{d(\cdot)}{dr} + \beta^{bx}(r) (\cdot)$ with
 138 $\alpha^{bx} = \frac{3}{r} - \frac{1}{E} \frac{dE}{dr}$ and $\beta^{bx} = (\nu - 1) \frac{1}{rE} \frac{dE}{dr}$.

139 Moreover, the hoop stress can be obtained from the equilibrium equation (2a)

$$\sigma_{\theta}^{bx}(r) = \sigma_r^{bx}(r) + r \frac{d\sigma_r^{bx}(r)}{dr}. \tag{7}$$

140 In parallel, and similar to the Kirsch solution, the pure shear problem can be solved
 141 by introducing the Airy stress function $\varphi(r, \theta)$ as follows [44]

$$\begin{aligned} \sigma_r^{ps}(r, \theta) &= \frac{1}{r} \frac{\partial \varphi(r, \theta)}{\partial r} + \frac{1}{r^2} \frac{\partial^2 \varphi(r, \theta)}{\partial \theta^2}, \\ \sigma_{\theta}^{ps}(r, \theta) &= \frac{\partial^2 \varphi(r, \theta)}{\partial r^2}, \\ \sigma_{r\theta}^{ps}(r, \theta) &= -\frac{\partial}{\partial r} \left(\frac{1}{r} \frac{\partial \varphi(r, \theta)}{\partial \theta} \right), \end{aligned} \tag{8a-c}$$

142 where φ has the form [44]

$$\varphi(r, \theta) = g(r) \cos 2\theta. \tag{9}$$

143 Consequently, Eqs. (8a-c) read

$$\begin{aligned} \sigma_r^{ps}(r, \theta) &= \left(\frac{1}{r} \frac{dg(r)}{dr} - \frac{4g(r)}{r^2} \right) \cos 2\theta, \\ \sigma_{\theta}^{ps}(r, \theta) &= \frac{d^2 g(r)}{dr^2} \cos 2\theta, \\ \sigma_{r\theta}^{ps}(r, \theta) &= 2 \left(\frac{1}{r} \frac{dg(r)}{dr} - \frac{g(r)}{r^2} \right) \sin 2\theta. \end{aligned} \tag{10a-c}$$

144 Combining Eqs. (10a-c) and (3a-c), the compatibility equation (4) reduces to the following
 145 differential equation

$$\mathcal{PS}(g(r)) = 0, \quad a \leq r < \infty \quad (11)$$

146 where the differential operator $\mathcal{PS}(\cdot)$ is given by $\frac{d^4(\cdot)}{dr^4} + \alpha^{ps}(r)\frac{d^3(\cdot)}{dr^3} + \beta^{ps}(r)\frac{d^2(\cdot)}{dr^2} +$

147 $\gamma^{ps}(r)\frac{d(\cdot)}{dr} + \delta^{ps}(r)(\cdot)$ with $\alpha^{ps} = \frac{2}{r} - \frac{2}{E} \frac{dE}{dr}$, $\beta^{ps} = -\frac{1}{E} \frac{d^2E}{dr^2} + \frac{2}{E^2} \left(\frac{dE}{dr}\right)^2 + \frac{\nu}{rE} \frac{dE}{dr} - \frac{2}{rE} \frac{dE}{dr} -$

148 $\frac{9}{r^2}$, $\gamma^{ps} = \frac{\nu}{rE} \frac{d^2E}{dr^2} - \frac{2\nu}{rE^2} \left(\frac{dE}{dr}\right)^2 + \frac{9}{r^2E} \frac{dE}{dr} + \frac{9}{r^3}$ and $\delta^{ps} = -\frac{4\nu}{r^2E} \frac{d^2E}{dr^2} + \frac{8\nu}{r^2E^2} \left(\frac{dE}{dr}\right)^2 - \frac{12}{r^3E} \frac{dE}{dr}$.

149 Relation (11) is a fourth-order linear differential equation with variable coefficients, and it
 150 is solved by considering the following boundary conditions

$$\sigma_r^{ps}(a, \theta) = 0,$$

$$\sigma_{r\theta}^{ps}(a, \theta) = 0,$$

$$\lim_{r \rightarrow \infty} \sigma_r^{ps}(r, \theta) = \frac{\sigma_0}{2} \cos 2\theta, \quad (12a-d)$$

$$\lim_{r \rightarrow \infty} \sigma_{r\theta}^{ps}(r, \theta) = -\frac{\sigma_0}{2} \sin 2\theta.$$

151 The set of the above equations for the two sub-problems can be found in [15].

152 3. The optimization problem: Motivation and formulation

153 Stresses for the case of a homogeneous infinite plate with a circular hole and subject
 154 to a uniaxial traction can be determined by taking Young's modulus as constant in the
 155 aforementioned equations, leading to the well-known Kirsch stress field [44]

$$\sigma_r(r, \theta) = \frac{\sigma_0}{2} \left(1 - \frac{a^2}{r^2}\right) + \frac{\sigma_0}{2} \left(1 + \frac{3a^4}{r^4} - \frac{4a^2}{r^2}\right) \cos 2\theta,$$

$$\sigma_\theta(r, \theta) = \frac{\sigma_0}{2} \left(1 + \frac{a^2}{r^2}\right) - \frac{\sigma_0}{2} \left(1 + \frac{3a^4}{r^4}\right) \cos 2\theta, \quad (13a-c)$$

$$\sigma_{r\theta}(r, \theta) = -\frac{\sigma_0}{2} \left(1 - \frac{3a^4}{r^4} + \frac{2a^2}{r^2}\right) \sin 2\theta.$$

156 It can be easily shown that the SCF at the rim of the circular hole is identically 3 by taking
 157 the limit of Eq. (13b) for $r \rightarrow a$ and $\theta = \pi/2$ and dividing by σ_0 . This value has been
 158 drastically reduced by replacing homogeneous materials by functionally graded ones. For
 159 instance, according to [15], one can reduce the SCF at the rim of the hole by suitably

160 varying the two heterogeneity factors n and β , linked to the Young's modulus through the
 161 relation

$$E(r) = E_\infty \left[1 + \beta \left(\frac{r}{a} \right)^n \right], \quad (14)$$

162 where $E_\infty = \lim_{r \rightarrow \infty} E(r)$, $-1 < \beta < 1$ and $n < 0$ (Figure 2a shows the radial distribution of
 163 Young's modulus for $n = -5$ and for different instances of $\beta < 0$). A similar relation for
 164 Poisson's ratio has been employed with different heterogeneity factors, but it was found
 165 that it does not affect stresses significantly (for this problem, the order of discrepancy is
 166 less than 1%). Expressions for the associated stress field on MN are lengthy and therefore
 167 omitted in this article, but represented in a graphical form in Figure 2b (see [15]). It is
 168 important to notice that although the SCF may arbitrarily tend to 0^+ , an increase of the hoop
 169 stress occurs elsewhere along the radius, say at $r = \tilde{a}$. Denoting here after by $\check{\sigma}_\theta(r)$ the
 170 hoop stress along the vertical line MN, such inevitable increase takes place as the improper
 171 integral $\lim_{r \rightarrow \infty} \int_a^r \check{\sigma}_\theta(t) dt$, resulting from the equilibrium between the applied load and
 172 occurring hoop stresses, is constant regardless of the Young's modulus distribution.

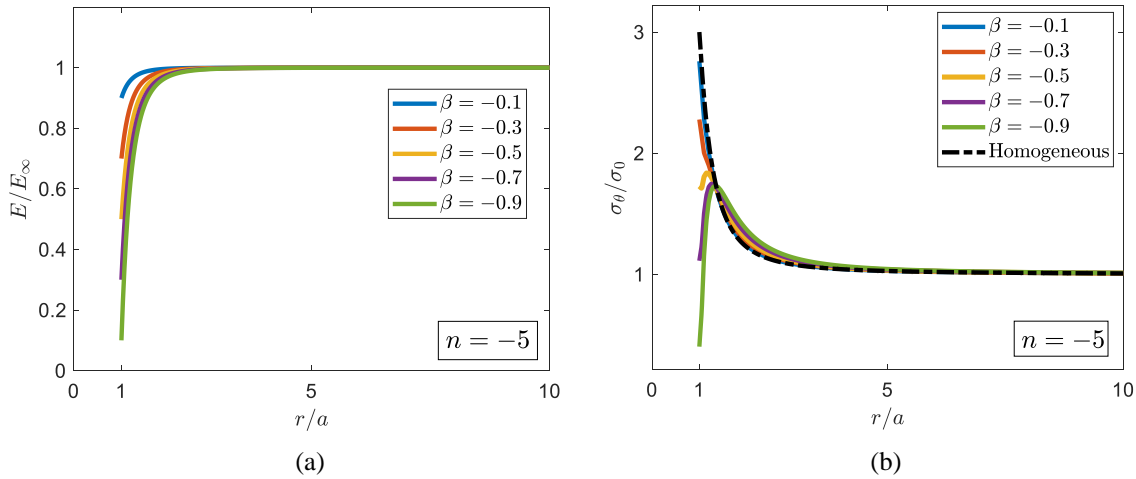


Fig. 2: (a) Variation of Young's modulus with r/a for $n = -5$ and for different values of $\beta < 0$. (b) The associated hoop stresses (solid lines) alongside with Kirsch solution (dashed line) on the vertical line MN. Stresses associated with other Young's modulus distributions are addressed in [15].

173 Thus, the optimum scenario, for the Young's modulus distribution (14), occurs when the
 174 heterogeneity factors lead to a constant hoop stress for $r \in [a, \tilde{a}]$, or, *lato sensu*, to a hoop
 175 stress whose standard deviation (or statistical variation) is as minimum as possible. This
 176 problem has not been addressed in [15], as authors focused on finding analytical solutions

177 for stresses. The formulation of the SCF minimization problem without remarkably
 178 increasing the hoop stress along the radius has been addressed in [16], albeit for a
 179 homogeneous isotropic infinite plate endowed with a functionally graded ring of radius
 180 $b > a$, where the Young's modulus distribution is given by

$$E(r) = E_b \left(\frac{r}{b}\right)^m, \quad (15)$$

181 where E_b is the Young's modulus at $r = b$ and m is a real positive number playing the role
 182 of the heterogeneity factor (see Figure 3a where different Young's modulus distributions
 183 are shown).

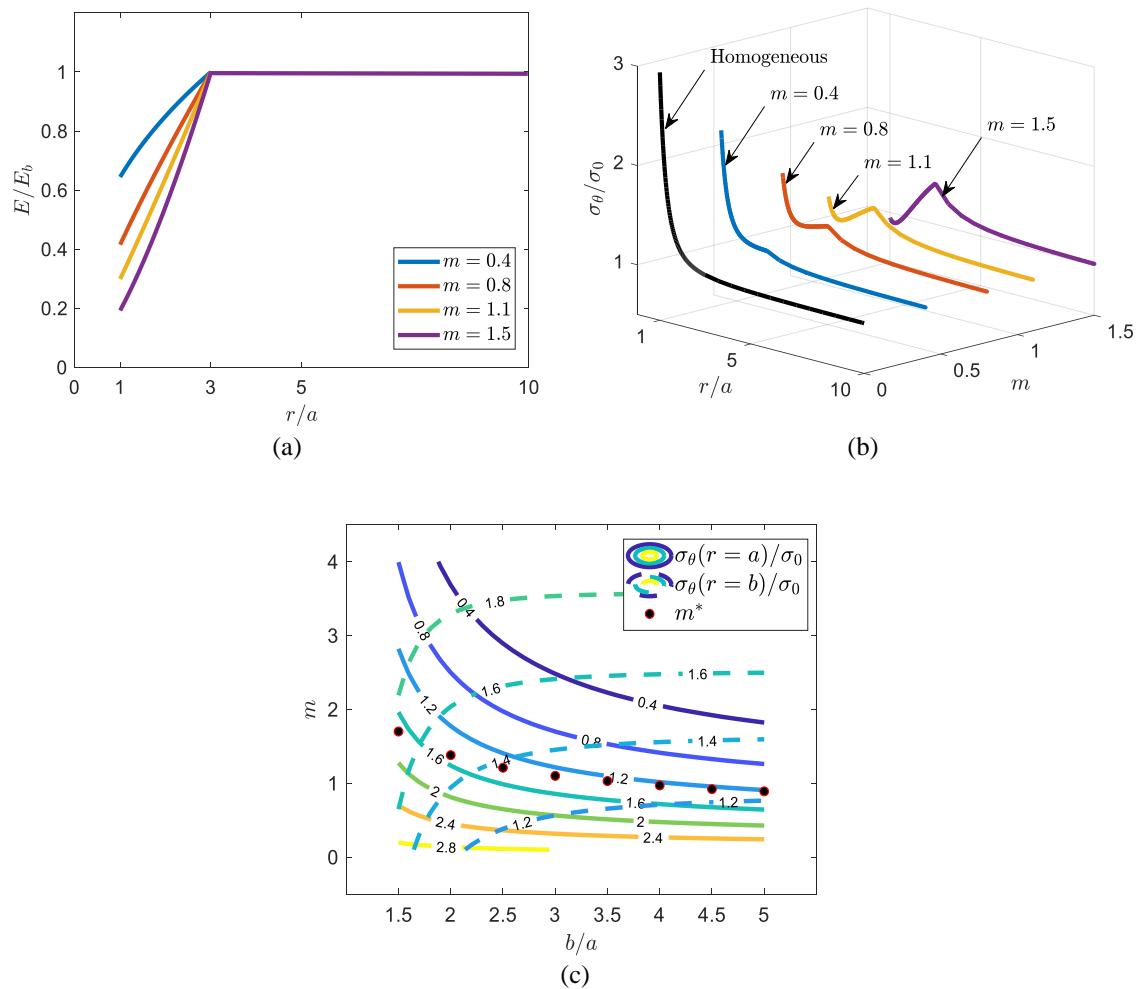


Fig. 3: (a) Variation of Young's modulus with $b/a = 3$ and for different values of $m > 0$. (b) The associated hoop stresses on the vertical line MN alongside with the Kirsch solution. (c) Contour levels for SCFs at the rim of the circular hole (solid contours), the interface between the ring and the homogeneous media (dashed contours) and best homogeneous factors m^* (scatter points). Stresses associated with other ring radii are addressed in [16].

184 Also here, Poisson's ratio ν is assumed constant and equal to the value of the homogeneous
 185 medium. Unlike [15], however, the author not only discussed the analytical tractability of
 186 the stress field (whose expression is omitted in this article), but also gave hints on the
 187 choice of the best heterogeneity factor for the optimum distribution for the hoop stress
 188 throughout the plate. In other words, the author showed that, regardless of the ring
 189 geometry, there exists a value of m , say m^* , such that the hoop stress assumes the same
 190 value at $r = a$ and $r = b$ and less elsewhere, provided that the search for m^* takes place
 191 in the range [16]

$$0 < m^* \leq \frac{8(2 - \sqrt{3})}{\nu - 7 + 4\sqrt{3}} \quad (16)$$

192 to avoid complex values for the stress field. Figure 3b shows the normalized hoop stresses
 193 along the vertical line MN for different instances of m and for $b/a = 3$ and compared with
 194 (13b). It is shown that the value of the best heterogeneity factor is approximately $m^* = 1.1$
 195 [16]. For completeness, it is desired to study the dependence of m^* on the geometry of the
 196 ring. A possible way is to compute contour levels for the SCFs at the rim of the circular
 197 hole and at the interface of the ring with the homogeneous medium for a range of
 198 admissible m , in the sense of the upper and lower limits given by Eq. (16), and for different
 199 values of b/a . By construction, the intersection of the two contour levels thus helps the
 200 reader identify the best heterogeneity factors m^* for fixed values of the ring geometry b/a .
 201 This practical chart is shown in Figure 3c, where the values for m^* are represented by
 202 scatter points. It is worth noting that the optimum heterogeneity factor monotonically
 203 decreases as b/a increases, namely a stiffer material at the circular hole is needed to
 204 compensate for the increase in the ring radius.

205 Based on the aforementioned considerations, an optimization problem in which the
 206 distribution of Young's modulus is sought for the minimization of the SCF arises. In order
 207 to avoid stress peaks along the radial direction the goal of minimizing the SCF can be
 208 replaced by the minimization of the maximum hoop stress along the line MN (see Figure
 209 1a), namely

$$\check{\sigma}_{\theta, \max} = \max_{r \geq a} \check{\sigma}_{\theta}(r), \quad (17)$$

210 as the hoop stress, for any (axisymmetric) Young's modulus variation, is expected to reach
 211 its peak only along this line. Hence, the optimization problem consists in finding the

212 Young's modulus distribution (or, through Eq. (1), the volume fraction) along the radial
213 direction such that the maximum value for the hoop stress along MN reaches its minimum
214 value, namely

$$\begin{aligned} \text{Problem 1.} \quad & \min_{V(r)} && (17), \\ & \text{s.t.} && (1), \\ & && (5a-c), \\ & && (6a-c), \\ & && (11), \\ & && (12a-d). \end{aligned}$$

215 Consequently, Problem 1 does not assume any a priori functional form of Young's modulus
216 along the radial direction.

217 In the parlance of optimization theory, Problem 1 is referred to as *dynamic*
218 optimization problem, namely an optimization problem whose decision variables are
219 unknown piecewise continuous functions living in a certain domain, and constraints are
220 differential relations. Solution to Problem 1 is cumbersome from the analytical viewpoint,
221 requiring one to resort to numerical methods. Among all, the so-called direct transcription
222 approach is used, which helps convert the dynamic optimization problem into a nonlinear
223 programming (NLP) problem, namely to an optimization problem whose decision
224 variables are collected in a finite-dimensional vector and constraints consist in equality or
225 inequality algebraic relations. The conversion of algebraic and differential constraints (5a-
226 c)-(6a-c) and (11)-(12a-d) into algebraic ones can be carried out by classic numerical
227 methods in mechanics such as the finite- element, volume, or difference methods. In this
228 article, the latter method is employed due to the simplicity of the boundary conditions of
229 the problem under consideration. Hence, the governing equations for the biaxial and pure
230 shear problems are solved by the finite difference method, which is recalled in the next
231 Section for the sake of a self-contained work. Subsequently, to validate the finite difference
232 code, an infinite functionally graded plate with a prefixed Young's modulus distribution of
233 the form (14) is numerically solved and compared to analytical solutions in [15].

234

235 **4. Direct transcription approach**

236 Hereinafter, the discretization scheme and matrices assembly are performed along
 237 the vertical line MN (i.e., with $\theta = \pi/2$) up to a limit radius A sufficiently large (namely
 238 $a \ll A < \infty$). Denoting by K the number of (equally distant) discretization points r_k ($k =$
 239 $1, 2, \dots, K$) and letting $r_1 = a$ and $r_k = A$, Table 1 lists the finite difference expressions
 240 employed to substitute the different derivatives appearing in the governing equations at the
 241 generic node r_k , where Ψ generically represents the unknown variable, i.e., either σ_r^{bx} in
 242 Eq. (6a) or g in Eq. (11), and $\Delta r = \frac{A-a}{K-1}$ denotes the radial step. Finite difference
 243 approximation terms have been chosen to guarantee a second-order accuracy.

Tab. 1: Second-order accuracy expressions for the finite difference terms for the approximation of the different derivatives [45]. Here, Ψ_k denotes the value of Ψ at the generic node r_k .

Node	Derivative	Approximation
First	Forward 1 st derivative	$\frac{d\Psi}{dr} \approx \frac{-\Psi_3 + 4\Psi_2 - 3\Psi_1}{2\Delta r}$
	Forward 2 nd derivative	$\frac{d^2\Psi}{dr^2} \approx \frac{2\Psi_1 - 5\Psi_2 + 4\Psi_3 - \Psi_4}{\Delta r^2}$
Last	Backward 1 st derivative	$\frac{d\Psi}{dr} \approx \frac{\Psi_{K-2} - 4\Psi_{K-1} + 3\Psi_K}{2\Delta r}$
	Backward 2 nd derivative	$\frac{d^2\Psi}{dr^2} \approx \frac{-\Psi_{K-3} + 4\Psi_{K-2} - 5\Psi_{K-1} + 2\Psi_K}{\Delta r^2}$
Intermediate	Central 1 st derivative	$\frac{d\Psi}{dr} \approx \frac{\Psi_{k+1} - \Psi_{k-1}}{2\Delta r}$
	Central 2 nd derivative	$\frac{d^2\Psi}{dr^2} \approx \frac{\Psi_{k+1} - 2\Psi_k + \Psi_{k-1}}{\Delta r^2}$
	Central 3 rd derivative	$\frac{d^3\Psi}{dr^3} \approx \frac{\Psi_{k+2} - 2\Psi_{k+1} + 2\Psi_{k-1} - \Psi_{k-2}}{2\Delta r^3}$
	Central 4 th derivative	$\frac{d^4\Psi}{dr^4} \approx \frac{\Psi_{k+2} - 4\Psi_{k+1} + 6\Psi_k - 4\Psi_{k-1} + \Psi_{k-2}}{\Delta r^4}$

244 Firstly, the finite difference method is applied to Eqs. (6a-c). Taking into account
 245 the different expressions in Table 1, and after some algebra, Eq. (6a) can be rewritten as
 246 the following system of $K - 2$ algebraic equations

$$\begin{aligned} \sigma_{r_{k+1}}^{bx} (2 + \Delta r \alpha_k^{bx}) + \sigma_{r_k}^{bx} (-4 + 2\Delta r^2 \beta_k^{bx}) \\ + \sigma_{r_{k-1}}^{bx} (2 - \Delta r \alpha_k^{bx}) = 0, \end{aligned} \quad k = 2, 3, \dots, K - 1 \quad (18)$$

247 while boundary conditions (6b,c) are simply replaced by their approximations

$$\begin{aligned} \sigma_{r_1}^{bx} &= 0, \\ \sigma_{r_K}^{bx} &= \frac{\sigma_0}{2}. \end{aligned} \quad (19a,b)$$

248 Equations (18)-(19a,b) can thus be written in the matrix form

$$\mathbf{A} \boldsymbol{\Sigma} = \mathbf{m}, \quad (20)$$

249 where $\mathbf{A} \in \mathbb{R}^{K \times K}$ is a square tridiagonal matrix, $\boldsymbol{\Sigma} \in \mathbb{R}^K$ is a column vector whose elements
 250 are the variables $\sigma_{r_1}^{bx}, \sigma_{r_2}^{bx}, \dots, \sigma_{r_K}^{bx}$ and $\mathbf{m} \in \mathbb{R}^K$ is a column vector whose first $K - 1$
 251 elements are zeros and the last one is $\sigma_0/2$.

252 The same considerations can be taken into account for Eqs. (11)-(12a-d). Equation
 253 (11) can be rewritten as the following system of $K - 4$ algebraic equations

$$\begin{aligned} g_{k+2} (2 + \Delta r \alpha_k^{ps}) + g_{k+1} (-8 - 2\Delta r \alpha_k^{ps} + 2\Delta r^2 \beta_k^{ps} + \Delta r^3 \gamma_k^{ps}) \\ + g_k (12 - 4\Delta r^2 \beta_k^{ps} + 2\Delta r^4 \delta_k^{ps}) \\ + g_{k-1} (-8 + 2\Delta r \alpha_k^{ps} + 2\Delta r^2 \beta_k^{ps} - \Delta r^3 \gamma_k^{ps}) \\ + g_{k-2} (2 - \Delta r \alpha_k^{ps}) = 0. \end{aligned} \quad k = 3, 4, \dots, K - 2 \quad (21)$$

254 Also here, the terms α_k^{ps} , β_k^{ps} , γ_k^{ps} and δ_k^{ps} can be derived by using the derivative
 255 approximations in Table 1 of their expressions. Finally, boundary conditions (12a-d), with
 256 the aid of Eqs. (10a-c), can be approximated as follows

$$\begin{aligned} \frac{1}{a} \frac{g_3 + 4g_2 - 3g_1}{2\Delta r} - \frac{4g_1}{a^2} &= 0, \\ \frac{1}{a} \frac{g_3 + 4g_2 - 3g_1}{2\Delta r} - \frac{g_1}{a^2} &= 0, \\ \frac{1}{A} \frac{g_{K-2} - 4g_{K-1} + 3g_K}{2\Delta r} - \frac{4g_K}{A^2} &= \frac{\sigma_0}{2}, \\ \frac{1}{A} \frac{g_{K-2} - 4g_{K-1} + 3g_K}{2\Delta r} - \frac{4g_K}{A^2} &= -\frac{\sigma_0}{4}, \end{aligned} \quad (22a-d)$$

257 respectively, where the far-field boundary conditions have been evaluated at the last
 258 discretization point $r_K = A \gg a = r_1$. The resulting system of equations can be recast in
 259 the matrix form

$$\mathbf{B} \boldsymbol{\Gamma} = \mathbf{n}, \quad (23)$$

260 where $\mathbf{B} \in \mathbb{R}^{K \times K}$ is the a square pentadiagonal matrix, $\boldsymbol{\Gamma} \in \mathbb{R}^K$ is a column vector whose
 261 elements are the variables g_1, g_2, \dots, g_K and $\mathbf{n} \in \mathbb{R}^K$ is a column vector whose first $K - 2$
 262 elements are zeros and the last two are $\sigma_0/2$ and $-\sigma_0/4$, respectively. Elastic uniform
 263 biaxial and pure shear stresses are embedded into Eqs. (20) and (23), respectively, whose
 264 solutions are given by

$$\boldsymbol{\Sigma} = \text{inv}(\mathbf{A}) \mathbf{m}, \quad (24)$$

265 and

$$\boldsymbol{\Gamma} = \text{inv}(\mathbf{B}) \mathbf{n}. \quad (25)$$

266 where $\text{inv}(\cdot)$ is the inverse operator for square matrices.

267 The finite difference method has been implemented successfully for the
 268 computation of stresses arising in functionally graded bodies in several circumstances, e.g.,
 269 [46,47]. Nevertheless, before proceeding with the solution of the optimization problem, an
 270 example showing the validation of the method is necessary. Analytical solutions for the
 271 stresses are thus borrowed from the literature and compared to the numerical results.
 272 Among others, closed-form solutions derived in [15] are taken into account, where
 273 mechanical properties are described by the general power-law (14) for $\beta = \pm 0.9$ and $n =$
 274 -5 . Figure 4a shows the analytical solutions for the radial and hoop stresses (solid lines)
 275 in the plate along MN and the numerical solutions (scatter points) by means of the finite
 276 difference method. A mesh convergence study has been carried out for the Young's
 277 modulus variation (14) adopted in [15] with $\beta = \pm 0.9$ and $n = -5$. In particular, it was
 278 found that the maximum values of the occurring stresses satisfy the convergence criterion
 279 $\sigma_{j,\max}^{(K_{i+1})} - \sigma_{j,\max}^{(K_i)} \leq 10^{-2} \text{MPa}$ beyond $K = 300$, being i and $i + 1$ two numerical forecasts
 280 employing K_i and K_{i+1} nodes, respectively, and $j = r, \theta$ (see Figure 4b).

281

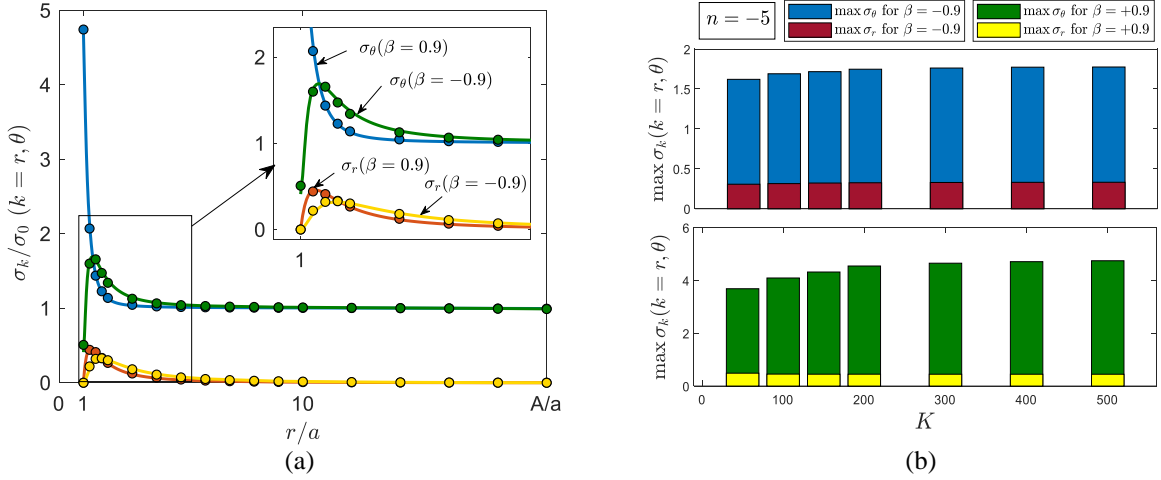


Fig. 4: (a) Analytical (solid lines) versus numerical (scatter points) solutions for the normalized radial and hoop stresses along the vertical line MN associated with the Young's modulus distribution in Eq. (14) with $\beta = \pm 0.9$ and $n = -5$. The parameters adopted for the simulation are $\nu = 0.3$ and $A/a = 20$. (b) Convergence study for the maximum radial and hoop stresses as functions of K .

282 Eventually, the maximum operator appearing in Eq. (17) is replaced by its p -norm
 283 approximation (where p is an even number greater than or equal 2), given by

$$\sigma_{\theta, \max} \approx \left(\int_a^A \check{\sigma}_\theta(r)^p dr \right)^{\frac{1}{p}} \quad (26)$$

284 and evaluated by means of the well-known trapezoidal rule, namely

$$\check{\sigma}_{\theta, \max} \approx \left[\Delta r \left(\check{\sigma}_\theta(a)^p + \check{\sigma}_\theta(A)^p + \sum_{i=2}^{K-1} \check{\sigma}_\theta(r_i)^p \right) \right]^{\frac{1}{p}}. \quad (27)$$

285 Thus, Problem 1 can be transcribed into the following NLP problem.

Problem 2. $\min_{\mathbf{V} \in \mathbb{R}^K}$ $\check{\sigma}_{\theta, \max} \approx \left[\Delta r (\check{\sigma}_{\theta 1}^p + \check{\sigma}_{\theta K}^p + \sum_{i=2}^{K-1} \check{\sigma}_{\theta i}^p) \right]^{\frac{1}{p}}$

s.t. $E_j = \tilde{E}_1(1 - V_j) + \tilde{E}_2 V_j, \quad j = 1, 2, \dots, K$

$\sum_{i=1}^K A_{ji} \sigma_{r i}^{bx} - m_j = 0, \quad j = 1, 2, \dots, K$

$\sum_{i=1}^K B_{ji} g_i - n_j = 0, \quad j = 1, 2, \dots, K$

$\check{\sigma}_{\theta j} - \sigma_{r j}^{bx} + r_j \frac{\sigma_{r j+1}^{bx} - \sigma_{r j-1}^{bx}}{2\Delta r} - \frac{g_{j+1} - 2g_j + g_{j-1}}{\Delta r^2} = 0, \quad j = 2, 3, \dots, K-1$

$\check{\sigma}_{\theta 1} - \sigma_{r 1}^{bx} + a \frac{-3\sigma_{r 1}^{bx} + 4\sigma_{r 2}^{bx} - \sigma_{r 3}^{bx}}{2\Delta r} - \frac{2g_1 - 5g_2 + 4g_3 - g_4}{\Delta r^2} = 0,$

$\check{\sigma}_{\theta K} - \sigma_{r K}^{bx} + A \frac{\sigma_{r K-2}^{bx} - 4\sigma_{r K-1}^{bx} + 3\sigma_{r K}^{bx}}{2\Delta r} - \frac{-g_{K-3} + 4g_{K-2} - 5g_{K-1} + 2g_K}{\Delta r^2} = 0,$

286 where the volume fraction has been replaced by a finite-dimensional vector $\mathbf{V} =$
 287 $(V_1, V_2, \dots, V_K) \in \mathbb{R}^K$, linked to Young's modulus through Eq. (1), being fixed the stiffness

288 ratio \tilde{E}_2/\tilde{E}_1 and the exponent p for the objective function evaluation. The vector of the
 289 decision variables of the NLP problem consists of the K discrete variables V_1, V_2, \dots, V_K .
 290 The constraints are the discrete equations for the elastic problems (6a-c) and (11)-(12a-d).
 291 The optimal solution therefore yields the optimal variation of the volume fraction and the
 292 corresponding stress behavior throughout the plate.

293 5. Results and discussion

294 In this Section, numerical optimal solutions for Problem 2 are illustrated and
 295 discussed. Hereinafter, the exponent p was taken to be equal to 200 (higher values
 296 generally lead to results too large to represent as conventional floating-point values during
 297 the iteration process), which yields a good approximation of the maximum hoop stress
 298 associated with the optimal solution, as confirmed by numerical results below. A gradient-
 299 based solver has been employed to numerically compute the optimal decision variable such
 300 that the maximum hoop stress reaches its minimum value. The algorithm used in this study
 301 is the well-known sequential quadratic programming algorithm [48]. Termination
 302 tolerances on both the function value as well as on the first-order condition for optimality
 303 have been imposed as 10^{-6} . In the light of conclusions made in [49], a linear volume
 304 fraction has been chosen as an initial guess and the numerical optimal solution has been
 305 sought iteratively.

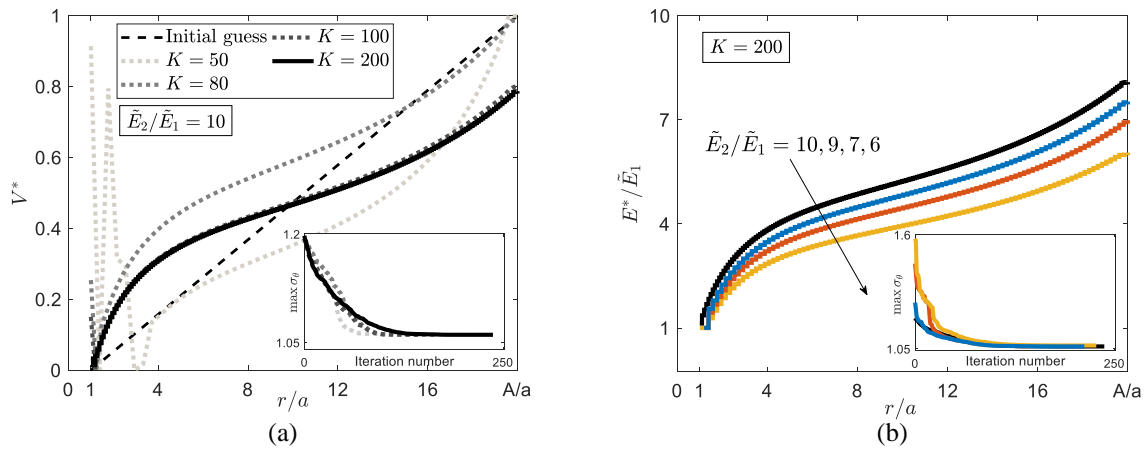


Fig. 5: (a) The linear initial guess (dashed line) as well as optimal numerical solutions (dotted and solid lines) for the volume fraction as K increases considering $E_2/E_1 = 10$. (b) Optimal solutions for the Young's modulus distribution considering different stiffness ratios. Numerical forecasts have been performed with $p = 200$ and $A/a = 20$. The history of the iterations is also reported in the lower-right angle of each figure.

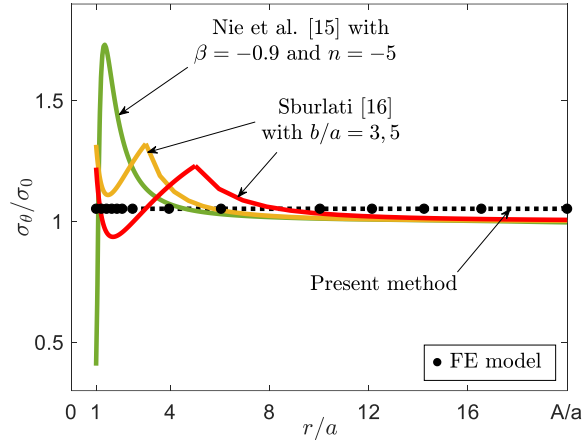
306 For the sake of comparison with the result obtained in [15], a first simulation has
307 been performed with a stiffness ratio $\tilde{E}_2/\tilde{E}_1 = 10$. The initial target is therefore to compare
308 numerical results with the stress performance associated with the Young's modulus
309 distribution obtained with $\beta = -0.9$ and $n = -5$ (see Figures 2a,b or 4). Figure 5a shows
310 the initial guess (dashed line) and numerical optimal volume fractions with the same load
311 and geometrical parameters as those employed for the validation example. More precisely,
312 successive numerical solutions were obtained for increasing K values (dotted lines) until a
313 prefixed convergence criterion between consecutive optimal solutions is achieved. In
314 particular, the considered optimal solution ($K = 200$, solid line) was chosen instead of
315 another ones associated with lower nodes (e.g., $K = 100$) as the norm of their difference
316 is less than 10^{-2} . It is worth noting that the optimal volume fraction increases throughout
317 the radial direction, indicating the optimality of adopting a softer material at the rim of the
318 circular hole. This finding is in agreement with the literature reporting enhancement studies
319 for the SCF for plates with circular holes (see, e.g., [10]). The resulting optimal Young's
320 modulus distribution is following a sort of sigmoid function around the linear distribution.
321 Moreover, the optimal material distribution does not necessarily assume, as base materials,
322 the functionally graded material constituents at the boundaries of the plate. Similar
323 forecasts have been performed for different stiffness ratios \tilde{E}_2/\tilde{E}_1 , leading to the same
324 conclusion (see Figure 5b).

325 To assess the stress performance of the optimal solution, the associated elastic
326 hoop, radial and shear stresses are respectively illustrated in Figures 6a,b,c (dotted lines).
327 It is worth appreciating that the hoop stress is uniform throughout the plate and free of
328 stress peaks, yielding a plateaued stress behavior and thus making the stress concentration
329 vanish throughout the radial domain. Moreover, the radial and shear stresses obey the
330 boundary conditions of the problem. It is worth noting that the optimization output is the
331 same if the uniaxial load direction is rotated by $\pi/2$, provided that the optimization
332 problem is formulated on the line associated with $\theta = 0$. To further assess the correctness
333 of the stress field obtained by the optimal Young's modulus distribution, a finite element
334 (FE) forecast was carried out by a commercial software (ANSYS Mechanical APDL 2022
335 R1). Due to symmetrical load and geometrical considerations, the geometrical domain
336 consists of the quarter of the plate and is discretized by means of second-order quadrilateral

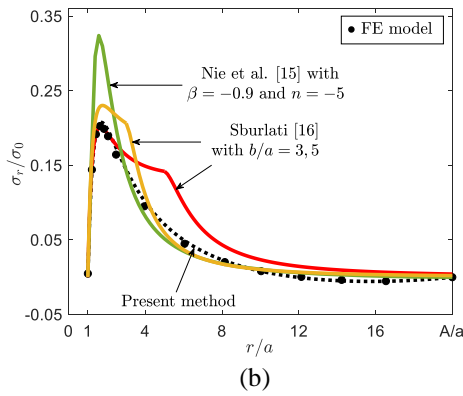
337 plane stress elements (PLANE 183). Necessary symmetric boundary conditions and the
338 uniaxial load have been suitably applied to the model. The radial direction has been
339 discretized into 200 radial strips (the same discretization points used in the transcription
340 procedure), each of which is isotropic and homogeneous and has the same mechanical
341 properties. Adjacent layers present different properties such that the resulting piecewise
342 constant variation approximates the optimal Young's modulus distribution displayed in
343 Figure 5b. The FE stress behavior has been represented by scatter points, showing a
344 remarkable fit with the optimization solver outputs, compared to the material modeling
345 simplifications necessary for the FE forecast, and confirming the uniformity of the hoop
346 stress on the line MN. Furthermore, a comparison between stresses obtained by the present
347 approach alongside with those analytically derived in [15] (associated with $\beta = -0.9$ and
348 $n = -5$) and in [16] for two different ring geometries ($b/a = 3$ and 5) is made (see solid
349 lines). It is clear that the present approach leads to Young's modulus distributions
350 (sigmoid-like curves) associated with the most uniform hoop stress (and consequently the
351 minimum peak hoop stress) throughout the plate.

352 **6. Conclusions**

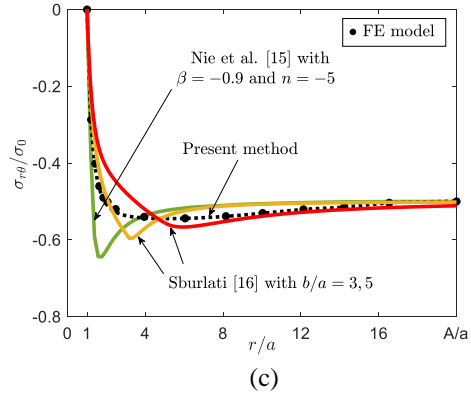
353 The optimization of the volume fraction distribution to minimize peak hoop stresses in
354 functionally graded infinite plates with a circular hole and subjected to uniaxial traction is
355 numerically addressed. The optimization problem has been stated and formulated as a
356 dynamic optimization problem, where the variation of the decision variable, namely the
357 volume fraction and consequently the Young's modulus through the rule of mixture, is
358 unknown beforehand and not limited to specified functions along the radial direction. The
359 problem has been divided into two sub-problems, i.e., occurring stresses have been
360 assumed as the superposition of those resulting from the biaxial and from the pure shear
361 deformations. Optimality conditions for the best distribution of the volume fraction could
362 not be solved analytically, hence numerical methods were necessary.



(a)



(b)



(c)

Fig. 6 Stresses associated with the optimal numerical solution (dotted lines) and finite element results (scatter points). (a) Hoop and (b) radial stresses along the vertical line MN and the (c) shear stress along $\theta = \pi/4$. Comparison between optimized stresses with results in literature (solid lines) [15,16].

363 The transcription procedure consisted in approximating the peak stress by the
 364 trapezoidal rule and converting the differential equations accounting for the elastic problem
 365 into two systems of algebraic equations describing the two sub-problems by means of the
 366 finite difference method due to the simplicity of the employed boundary conditions which
 367 permitted the solution with reduced computational costs. The obtained numerical solutions
 368 for the Young's modulus follow a sigmoidal behavior. The associated hoop stress revealed
 369 uniform along the radius and has been validated by the finite element method.

370 The Young's modulus distribution has been assumed to follow the rule of mixture;
 371 however, other models for the effective Young's modulus can be fitted in the same
 372 framework. The present article can be considered as a preliminary study whose results can
 373 be further extended as follows. For instance, other geometrical discontinuities such as

374 elliptic or rounded-square holes can be taken into account, provided that the transcription
375 of the differential elastic equations into algebraic equations is replaced by a suitable finite
376 element approach.

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